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A Control Oriented Model with Variable Valve Timing for HCCI Combustion Timing Control

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ABSTRACT

Homogeneous Charge Compression Ignition (HCCI) is a promising concept for combustion engines to reduce both emissions and fuel consumption. HCCI combustion control is a challenging issue because there is no direct initiator of combustion. Variable Valve Timing (VVT) is being used in SI engines to improve engine efficiency. When VVT is used in conjunction with HCCI combustion it is an effective way to control the start of combustion. VVT changes the amount of trapped residual gas and the effective compression ratio for each cycle both of which have a strong effect on combustion timing in HCCI engines. To control HCCI combustion, a physics based control oriented model is developed that includes the effect of trapped residual gas on combustion timing. The control oriented model is obtained by taking a physics based model of the reaction kinetics and transient dynamics and systematically reducing the model using simplification of reaction mechanisms. This method allows different fuels to be incorporated using a standard methodology. The reduced order model consists of these five stages: intake, compression, combustion, expansion and exhaust. This model fills the gap between complex models with highly detailed chemical kinetics and simple black box dynamic models that have been used in model based control.

INTRODUCTION

Homogeneous charge compression ignition (HCCI) engines have the potential for high efficiencies and low pollutant emissions. HCCI engines are a combination of both conventional Spark Ignition (SI) and diesel Compression Ignition (CI) engines. In HCCI engines, fuel and air are homogeneously mixed and compressed similar to SI engines. When the piston is near to top dead center of the compression stroke, the air-fuel mixture auto-ignites at many locations. In order to control the rate of pressure rise in HCCI engines, the

air and fuel mixture is diluted with burned gases. The combination of a diluted and premixed air fuel mixture with multiple ignition points inside the combustion chamber provides low temperature combustion zones that reduce the production of nitrogen oxides [1].

HCCI engines have no direct in-cylinder mechanism for combustion initiation, unlike the ignition event in SI and CI engines which are controlled by spark and fuel injection respectively. There are several ways to control the combustion timing in an HCCI engine. Intake air heating is one effective way for combustion timing control [2]. Usually, an electric air heater is used to raise the intake air temperature. This approach is not efficient because energy is required to heat the air and the heater response time is long compare to an engine cycle. Exhaust Gas Recirculation (EGR) is another way to control combustion timing. EGR has four effects [3, 4] but mainly increases the mixture temperature and dilutes the premixed charge [5, 6, 7, 8, 9, 10, 11, 12]. The combustion products can be retained or reinducted using Variable Valve Timing (VVT). This reduces heat loss and achieves fast control response compared to external EGR. Variable compression ratio engines can also be used to control the combustion timing [13, 14] but the mechanism is complex. Controlling the combustion timing by varying the autoignition properties of the fuel using dual-fuels is also effective but two fuels are needed [15, 16, 17]. Direct fuel injection is another way that can be used to control the auto ignition properties of the premixed charge [18, 19]. Water injection has also been used to cool the mixture and control the combustion timing in HCCI engines since the ignition timing is very temperature sensitive [20, 21].

HCCI combustion timing is controlled purely by chemical kinetics of the trapped charge [1, 22]. Therefore, to achieve ignition timing control of HCCI combustion, the mixture

composition, temperature and pressure at the Inlet Valve Closing (IVC) must be controlled. VVT enables quick changes in the amount of trapped hot residual gases inside the cylinder and is a simple and precise actuation method to set conditions at IVC for desired auto-ignition timing.

To implement closed-loop HCCI combustion timing control, a control oriented model is needed. Modeling has been found to be an important part of HCCI engine controller design. There are two main methods to obtain an HCCI control oriented model: physical [23, 24, 25] and system identification [15, 16, 26]. Physical models are classified according to the number of spatial dimensions in the cylinder, e.g. zero-dimensional [27, 28, 29, 30, 31] or computational fluid dynamics (three dimensional) models [32, 33, 34, 35]. Zero dimensional models provide no spatial resolution. Most of the zero-dimensional models are based on detailed chemical kinetics and are not suitable for control analysis. Computational fluid dynamics models are even more complex and usually produce only a single engine cycle of data. For control analysis, a model that is capable of quickly simulating many engine cycles is required.

Many HCCI engine models for control purposes have been developed [36, 37, 38]. In these models, the combustion mechanism is usually greatly simplified to decrease the simulation time. Ohyama [36] developed a real time physics based model for in-cylinder Air/Fuel ratio and trapped residual gas mass fraction estimation where air and residual gas mass fraction at the beginning of compression were determined based on signals from an air flow meter and in-cylinder pressure transducer. Shaver et. al. [37] developed a physical based control oriented model for a propane HCCI engine. An integrated Arrhenius rate expression was used to capture the importance of species concentrations and temperature on the ignition process. Rausen et. al. [39] developed a mean value model for the control of a HCCI engine. Their model has five continuous and three discrete states and the effects of exhaust gas recirculation and re-breathing was investigated. They used a simple Arrhenius integral model to estimate the start of combustion and algebraic equations were used for the heat release rate calculation. Ognik et. al. [38] developed a physics based model for a gasoline HCCI engine considering variable valve timing. Their model connected the SENKIN code of the CHEMKIN library [40] to the AVL BOOST [41] engine cycle simulation code and parametric studies of the combustion process in a single cylinder HCCI engine were described. Kara-giorgis et. al. [42] developed a simple non-linear low-order control oriented physical based model for a gasoline HCCI engine which is suitable for variable valve actuated HCCI engines. They used simple thermodynamics concepts to predict in-cylinder pressure and temperature at IVC. The gas exchange was based on in-cylinder dynamics and it was assumed that the manifold pressure is constant during gas exchange. Combustion was modeled in a semi

empirical fashion. In [16], a system identification approach was used on a dual-fuel HCCI engine to generate input-output models for control. In [43], a control oriented model that simulates the engine cycle from the intake to the exhaust stroke including the thermal coupling dynamics caused by the residual gases from one cycle to the next cycle was developed. In their work, the gas exchange process, engine output work and combustion were predicted using semi-empirical correlations.

Despite extensive work on the modeling of HCCI combustion, to date no physical control oriented model can provide accurate and fast prediction of the combustion timing with the variation of trapped residual gas. Thus the focus of this work is to develop a control oriented model of HCCI combustion timing using a detailed physical based model.

DETAILED PHYSICAL MODEL **(DPM)**

A detailed HCCI four stroke cycle is modeled as a sequence of continuous processes: intake, compression, combustion, expansion and exhaust. In the cycle simulation, the system of interest is the instantaneous contents of a cylinder. This system is open to the transfer of mass, enthalpy and energy in the form of work and heat. The cylinder is modeled as a time variant volume and the cylinder contents are divided into fourteen continuous zones. Quasi steady, adiabatic, one dimensional flow equations are used to predict the mass flow past the intake and exhaust valves. The intake and exhaust manifolds are modeled as constant volumes whose pressure and temperature are determined by solving mass and energy equations for each manifold. Intake charge and exhaust gas are modeled as ideal gases. A reduced order reaction mechanism for n-heptane is used for combustion simulation. The reaction mechanism is from [44]. This reduced mechanism consists of 29 species and 52 reactions and is generated from the detailed n-heptane reaction mechanism [45].

Conservation of mass is used to develop a differential equation for the change in species concentration and energy conservation is used to obtain a differential equation for the change in system temperature. For the in-cylinder content, conservation of energy can be written [46] as:

$$\dot{U} = \sum_j \dot{m}_j h_j + \dot{Q}_W - \dot{W} \quad (1)$$

where U is the internal energy, \dot{Q}_W is the heat transfer rate into the system, \dot{W} is the rate at which the system does work by boundary displacement and h_j is the enthalpy of the j th specie entering or leaving the system. The internal energy is calculated as the sum of the internal energies of all species

$$U = \sum_k m_k u_k \quad (2)$$

where k represents the zone number. Differentiating Equation 2 with respect to time gives

$$\dot{U} = \sum_k m_k \dot{u}_k + \sum_k \dot{m}_k u_k \quad (3)$$

For an ideal gas the change in internal energy can be written as:

$$\dot{u}_k = \bar{c}_v \dot{T}_k \quad (4)$$

The conservation of gas species in each zone can be calculated as:

$$\dot{y}_{k,j} = \frac{\dot{\omega}_{k,j} M_j}{\rho} + \sum_j \frac{\dot{m}_j}{m_{cyl}} (y_j - y_{cyl}) \quad (5)$$

where $\dot{\omega}_j$ is the net chemical production rate for each specie and M_j is the molar mass of each specie. Cantera [47] is used for the calculation of the net chemical production rate, internal energy and enthalpy of the gas species. Cantera is an open-source chemical kinetics software. The rate of change of mass in each zone is calculated as

$$\dot{m}_k = m_k \dot{y}_{k,j} \quad (6)$$

\dot{W} in Equation 1 is calculated as

$$\dot{W} = P_{cyl} \dot{V}_k \quad (7)$$

where \dot{V}_k and P_{cyl} are obtained from ideal gas law [27]

$$P_{cyl} = \frac{\sum_k m_k R_k T_k}{V_{cyl}} \quad (8)$$

$$\dot{V}_k = \dot{V}_{cyl} \frac{m_k R_k T_k}{(\sum_k m_k R_k T_k)^2} (m_k R_k \dot{T}_k \sum_k m_k R_k T_k - m_k R_k T_k \sum_k m_k R_k \dot{T}_k) \quad (9)$$

The cylinder volume, as a function of engine crank angle θ , are calculated using the slider crank mechanism [48]

$$V_{cyl} = V_c + \frac{\pi B^2}{4} [l + a - a \cos \theta - \sqrt{l^2 - (a \sin \theta)^2}] \quad (10)$$

where V_c is the cylinder clearance volume, B is the cylinder bore, l is the connecting rod length and $L = 2a$ is the stroke.

The rate of change of temperature in each zone can be expressed by substituting Equations 3, 4, 5, 6, 7, 8, 9 into Equation 1, resulting in:

$$\dot{T}_k = \frac{1}{\bar{c}_p} \left(\sum_j \dot{m}_j h_j + \dot{Q} - \sum_j u_j \dot{y}_{k,j} + \frac{R_k T_k}{\sum_k m_k R_k T_k} \sum_k m_k R_k \dot{T}_k - \frac{R_k T_k}{V_{cyl}} \dot{V}_{cyl} \right) \quad (11)$$

The heat transfer rate between the cylinder wall and the adjacent zone is calculated using a convective heat transfer law. A slightly modified version of Woschni's correlation has shown good results for HCCI engines and is used here to determine the heat transfer coefficient [48]:

$$\dot{Q} = A_W h_c (T_{cyl} - T_{wall}) \quad (12)$$

$$h_c = 3.26 B^{-0.2} p_{cyl}^{0.8} T_{cyl}^{-0.55} \bar{n}^{0.8} \quad (13)$$

$$\bar{n} = C_1 \bar{S}_p \quad (14)$$

where A_W is the cylinder wall area available for heat transfer, h_c is the heat transfer coefficient, and T_{wall} is the in-cylinder wall surface temperature. C_1 is 6.18 during induction and exhaust and 2.28 otherwise. B is the cylinder bore and \bar{S}_p is the mean piston speed. The heat transfer between each zones due to the temperature difference is calculated based on a simple conduction model developed in [27]. Residual gas mass and temperature distribution at IVC are obtained by the model proposed in [27].

Governing equations describing the exhaust manifold dynamics are similar to those describing HCCI combustion dynamics. The following two coupled differential equations describe exhaust manifold dynamics [49]:

$$\dot{m}_{man} = \dot{m}_{in} - \dot{m}_{out} \quad (15)$$

$$\dot{U} = \sum_j m_j \dot{h}_j + \dot{Q} \quad (16)$$

where $U = m_{man} c_v T_{man} = \frac{1}{\gamma-1} p_{man} V_{man}$ and $h_j = c_{p,j} T_j$.

Equations 15 and 16 are coupled with the ideal gas law:

$$p_{man} V_{man} = m_{man} R T_{man} \quad (17)$$

Rearranging Equations 15 and 16 using Equation 17 and neglecting heat transfer gives:

$$\dot{p}_{man} = \frac{\gamma R}{V_{man}} (\dot{m}_{in} T_{in} - \dot{m}_{out} T_{man}) \quad (18)$$

$$\begin{aligned} \dot{T}_{man} = & \frac{T_{man} R}{p_{man} V_{man} c_v} (\bar{c}_p \dot{m}_{in} T_{in} \\ & - \bar{c}_p \dot{m}_{out} T_{man} - \bar{c}_v (\dot{m}_{in} - \dot{m}_{out}) T_{man}) \end{aligned} \quad (19)$$

where γ is the heat capacity ratio and c_p and c_v are the specific heat at constant pressure and volume respectively. The conservation of manifold gas species can be expressed as:

$$\dot{y}_{man} = \sum_j \frac{\dot{m}_j}{m_{man}} (y_j - y_{man}) \quad (20)$$

The adiabatic formulations 18, 19 and 20 estimate the exhaust manifold behavior.

In this study a heater is used to increase the intake manifold temperature to a constant, hence an isothermal assumption is used for intake manifold modeling. Equations 18 and 19 can be simplified [49] to:

$$\dot{p}_{man} = \frac{RT_{man}}{V_{man}} (\dot{m}_{in} - \dot{m}_{out}) \quad (21)$$

$$T_{man} = \text{constant} \quad (22)$$

The equation for the conservation of the intake charge species is similar to Equation 20 so it is not repeated here. A one dimensional quasi steady compressible flow model is used to calculate mass flow rates through the intake and exhaust valves during induction and exhaust strokes. The intake and exhaust manifolds are treated as volumes with known pressure, temperature and mixture composition. When reverse flow to the manifolds occurs, a rapid mixing model is used. At each step of the intake or the exhaust strokes, values for the valve open areas are obtained from tabulated data of the actual valve profile. Given the valve open area, the discharge coefficient, and the pressure ratio across a valve, the mass flow rate across the valve is calculated [46] from:

$$\begin{aligned} \dot{m} = & C_d A \frac{P_o}{RT_o} \\ & \sqrt{\gamma RT_o} \left(\frac{2}{\gamma - 1} \left[\left(\frac{P_s}{P_o} \right)^{\frac{2}{\gamma}} - \left(\frac{P_s}{P_o} \right)^{\frac{\gamma+1}{\gamma}} \right] \right)^{\frac{1}{2}} \end{aligned} \quad (23)$$

where C_d is the discharge coefficient, A is the valve open area, P_o is the pressure upstream of the valve, P_s is the pressure downstream of the valve, T_o is the temperature upstream of the valve, γ is the ratio of specific heats and R is

the gas constant. For the case of choked flow, Equation 23 reduces [46] to:

$$\dot{m} = C_d A \frac{P_o}{RT_o} \sqrt{\gamma RT_o} \left(\frac{2}{\gamma + 1} \right)^{\frac{(\gamma+1)}{2(\gamma-1)}} \quad (24)$$

The throttle is considered as a flow restriction area and Equations 23 and 24 are applied for throttle body simulation.

Engine speed is modeled using a constant engine inertia. The differential equation for engine speed is:

$$I_{eng} \dot{\Omega} = T_{eng} - T_{load} \quad (25)$$

where T_{eng} is engine torque, Ω is engine speed, T_{load} is load torque and I_{eng} is the engine inertia [49].

The state equations for the model are given by Equations 5, 8, 11, 18, 19, 20, 21 and 25. The system under study can be written in the following general form, where x is the state of the system, u is the control input, w is a disturbance, and f is a nonlinear function.

$$\dot{x} = f(t, x, u, w) \quad (26)$$

where $x \in \mathbb{R}^n$, $u \in \mathbb{R}^m$ and $w \in \mathbb{R}^p$. This notation is a vector notation, which allows us to represent the system in a compact form. The main inputs of the model are the intake manifold temperature, valve timing, fuel mass flow rate and the load torque, $u^T = [\dot{m}_f \ T_{int} \ \theta_{IVO} \ \theta_{EVC} \ T_{load}]$. In addition to the inputs, the model also includes certain output variables that can be used to monitor and control the system. These outputs are combustion timing, peak pressure, pressure rise rate and output work, $y^T = [\theta_{SOC} \ P_{max} \ PRR \ W]$. The state variables are chosen due to their physical significance to the combustion process and are: in-cylinder and manifold temperature and pressure, species mass fraction and engine speed, $x^T = [T_{cyl} \ P_{cyl} \ T_{man} \ P_{man} \ y_{man} \ y_{cyl} \ \Omega]$. So for the detailed physical model $n=483$ [14(zones)×29(species in each zone)+14(temperature at each zone)+1 (in-cylinder pressure)+2(manifolds)×29(species)+2(intake and exhaust manifolds temperatures)+2(intake and exhaust manifolds pressures)] and $m=5$.

CONTROL ORIENTED MODEL (COM)

Model reduction of the previously described DPM model is used to develop a Control Oriented Model (COM) of a single cylinder HCCI engine. COM is developed based on the approach proposed in [50]. The system of interest is the instantaneous contents of the cylinder. It is assumed that there is no spatial variation in properties within a cylinder at any

instant of time. In order to determine the thermodynamic state of the mixture at IVC, the contents of the in-cylinder mixture must first be determined. The contents of the cylinder at the current cycle k depend on fresh reactants inducted during cycle k and residual gases from the last cycle, $k - 1$. For lean combustion of n-heptane with internal EGR, the inducted charge becomes:

$$\phi_k C_7 H_{16} + c_1 (\alpha_{i,k} (c_2 - \phi_{k-1}) + c_2) O_2 + c_3 (c_2 + \alpha_{i,k}) N_2 + c_4 \alpha_{i,k} \phi_{k-1} CO_2 + c_5 \alpha_{i,k} \phi_{k-1} H_2 O \quad (27)$$

where α_i , ϕ and k represents the residual gas fraction, fuel equivalence ratio and cycle number respectively and c_i are constants listed in Table 1. The values of α_i are obtained from the formation of the energy balance for the cylinder gas over the intake process from IVO to IVC. The fuel equivalence ratio and residual gas fraction are calculated [50, 51] as:

$$\phi_k = \frac{n_{f,k}}{n_{fs,k}} \quad (28)$$

$$\alpha_{i,k} = \frac{A P_{EVO,k-1}}{A P_{EVO,k-1} + B T_{EVO,k-1}} \quad (29)$$

and n_{fk} is the number of fuel moles, $n_{fs,k}$ is the number of fuel moles for stoichiometric combustion, $P_{EVO,k-1}$ is the pressure at exhaust valve open, and $T_{EVO,k-1}$ is the temperature at exhaust valve open. The factors A and B are:

$$A = V_{EVC,k-1} \Omega M_f T_{int,k} M_{air} \\ B = M_{res} P_{int,k} (V_{IVC,k} - V_{IVO,k}) \Omega M_{fuel} + c_6 R M_{res} (m_{f,k} - m_{f,k-1}) T_{int,k} M_{air}$$

$V_{IVC,k}$ is the in-cylinder volume at IVC, $V_{IVO,k}$ is the in-cylinder volume at inlet valves open, and $T_{int,k}$ is the intake charge temperature.

Temperature at IVC, T_{IVC} , is the temperature of the mixture at the time of inlet valve closing and is calculated as

$$T_{IVC,k} = \frac{C_{2,k} \alpha_{i,k} T_{EVO,k-1} + C_{1,k} T_{int,k}}{C_{1,k} + C_{2,k} \alpha_{i,k}} \quad (30)$$

where

$$C_{1,k} = \phi_k C_{p,C_7 H_{16}} + c_1 C_{p,O_2} + c_3 C_{p,N_2} \\ C_{2,k} = c_4 \phi_{k-1} C_{p,CO_2} + c_5 \phi_{k-1} C_{p,H_2 O} \\ + c_3 C_{p,N_2} + c_1 (c_2 - \phi_{k-1}) C_{p,O_2}$$

The intake process is assumed to take place at atmospheric pressure as the variable valve timing engine usually operates at wide-open throttle with the valve timing controlling the amount of air.

$$P_{IVC,k} = P_{int,k} \quad (31)$$

The compression of in-cylinder gas made up of fuel, air and residual gas is assumed to be isentropic. This results in the following equations:

$$T_{SOC,k} = \left(\frac{V_{IVC,k}}{V_{SOC,k}} \right)^{\gamma_c - 1} T_{IVC,k} \quad (32)$$

$$P_{SOC,k} = \left(\frac{V_{IVC,k}}{V_{SOC,k}} \right)^{\gamma_c} P_{IVC,k} \quad (33)$$

where γ_c is determined from experimental data. The crank angle of start of combustion is calculated using a simplified Arrhenius equation [50] as:

$$\theta_{SOC,k} = \theta_{IVC} + \frac{K_{th} \Omega}{A} e^{\left(\frac{E_a}{RT_{IVC,k} \left(\frac{V_{IVC,k}}{V_{TDC}} \right)^{\gamma - 1}} \right)} \\ \frac{(\alpha_{i,k} (c_7 \phi_{k-1} + c_8) + c_7 \phi_k + c_8)}{c_1^b (\phi_k)^a (\alpha_{i,k} (c_2 - \phi_{k-1}) + c_2)^b} \\ \left(\frac{RT_{IVC,k} V_{TDC}}{P_{IVC,k} V_{IVC,k}} \right)^{a+b} + \theta_{offset} \quad (34)$$

where the values of a , b , A , θ_{offset} , E_a and K_{th} are listed in Table 1.

Based on the correlations developed in [50] and [52] combustion duration, $\Delta\theta$, is calculated as:

$$\Delta\theta_k = c_9 c_{10}^{\phi_k} c_{11} T_{IVC,k} \left(\frac{V_{IVC,k}}{V_{SOC,k}} \right)^{\gamma - 1} c_{12}^{\theta_{SOC,k}} \quad (35)$$

Finally, θ_{50} , the crank angle of fifty percent fuel mass fraction burned, is calculated as:

$$\theta_{50,k} = \theta_{SOC,k} + 0.5 \Delta\theta_k \quad (36)$$

The first law of thermodynamics is applied to the system to determine the thermodynamic state of the system after combustion. In-cylinder gas temperature after combustion is calculated as:

$$T_{AC,k} = \frac{D_k + (D_{1,k} - N_{1,k} R) T_{2,k} - D_{1,k} c_{14} + D_{2,k} c_{14}}{D_{2,k} - R N_{2,k}} \quad (37)$$

where

$$\begin{aligned}
 D_k &= c_{13}LHV_{C_7H_{16}}\phi_k \\
 D_{1,k} &= \phi_k C_{p,C_7H_{16}} + c_1(\alpha_{i,k}(c_2 - \phi_{k-1}) + c_2)C_{p,O_2} + \\
 &\quad c_3(c_2 + \alpha_{i,k})C_{p,N_2} + c_4\alpha_{i,k}\phi_{k-1}C_{p,O_2} + c_5\alpha_{i,k}\phi_{k-1}C_{p,H_2O} \\
 D_{2,k} &= c_5(\phi_k + \alpha_{i,k}\phi_{k-1})C_{p,H_2O} + c_4(\phi_k + \alpha_{i,k}\phi_{k-1})C_{p,CO_2} + \\
 &\quad c_3(c_2 + \alpha_{i,k})C_{p,N_2} + c_1(c_2 + \alpha_{i,k} - \alpha_{i,k}\phi_{k-1} - \phi_k)C_{p,O_2} \\
 N_{1,k} &= \phi_k + c_1(\alpha_{i,k}(c_2 - \phi_{k-1}) + c_2) + c_3(c_2 + \alpha_{i,k}) + \\
 &\quad c_4\alpha_{i,k}\phi_{k-1} + c_5\alpha_{i,k}\phi_{k-1} \\
 N_{2,k} &= c_5(\phi_k + \alpha_{i,k}\phi_{k-1}) + c_4(\phi_k + \alpha_{i,k}\phi_{k-1}) + c_3(c_2 + \alpha_{i,k}) + \\
 &\quad c_1(c_2 + \alpha_{i,k} - \alpha_{i,k}\phi_{k-1} - \phi_k)
 \end{aligned}$$

The in-cylinder pressure after combustion is calculated by applying the ideal gas law before and after combustion as:

$$P_{AC} = \frac{N_{2,k}}{N_{1,k}} P_{SOC,k} \frac{T_{AC,k}}{T_{SOC,k}} \quad (38)$$

The rate of pressure rise is calculated from

$$PRR_k = \frac{P_{AC,k} - P_{SOC,k}}{\Delta\theta_k} \quad (39)$$

Rate of pressure rise is required for proper engine control. Large amounts of energy are released during a short period of time in HCCI engines. For this reason, a pressure rise rate threshold is usually determined to keep the combustion noise under a certain level.

The expansion process, which takes place until the opening of the exhaust valve, is assumed to be isentropic. The temperature and pressure of the in-cylinder gas at EVO are calculated as

$$T_{EVO,k} = \left(\frac{V_{AC,k}}{V_{EVO,k}} \right)^{\gamma_e - 1} T_{AC,k} \quad (40)$$

$$P_{EVO,k} = \left(\frac{V_{AC,k}}{V_{EVO,k}} \right)^{\gamma_e} P_{AC,k} \quad (41)$$

Values of all model parameters and constants are listed in Table 1. Those model parameters are determined from the DPM: γ , K_{th} , c_{13} and θ_{offset} . The parameters γ_c and γ_e are determined from experimental data. The parameters: a , b , A and E_a are taken directly from [50, 53].

Rewriting the COM in standard form, where x is the state of the system, u is the control input and w is a disturbance.

$$x_{k+1} = f(x_k, u_k, w_k) \quad (42)$$

$$y_k = g(x_k, u_k, w_k) \quad (43)$$

results in the input, the state, the output and the disturbance as:

$$u = [\theta_{IVO} \quad \theta_{EVC} \quad \dot{m}_f] \quad (44)$$

$$x = [\alpha_i \quad T_{IVC} \quad \theta_{SOC}] \quad (45)$$

$$w = [\Omega \quad \phi \quad T_{int} \quad P_{EVO}] \quad (46)$$

$$y = \theta_{50} \quad (47)$$

Table 1. Model parameters and constants

| | | | |
|-------------------|-----------------------|----------|------------------------|
| a | 0.25 | c_3 | 41.4 |
| b | 1.5 | c_4 | 7 |
| E_a | 15098 | c_5 | 8 |
| R | 0.8314 | c_6 | 2 |
| K_{th} | 2.31×10^{-6} | c_7 | 4 |
| γ | 1.37 | c_8 | 52.4 |
| γ_e | 1.34 | c_9 | 2.07×10^{-18} |
| γ_c | 1.32 | c_{10} | 3.55 |
| A | 5.1×10^{11} | c_{11} | 0.993 |
| θ_{offset} | 85.2 | c_{12} | 1.16 |
| c_1 | 11 | c_{13} | 0.87 |
| c_2 | 1 | c_{14} | 298 |

DPM AND COM VALIDATION

Both the DPM and COM are validated against experimental data from a single cylinder research engine with the specifications listed in Table 2. The simulations are conducted for a range of equivalence ratios and internal EGR levels. The test points that are used for model validation are listed in Table 3.

Figures 1, 2, 3, 4, 5 compare modeled cylinder pressure profiles and the corresponding 300 cycles of averaged experimental cylinder pressure traces. Both COM and DPM match the experimental pressure trace during compression, combustion and expansion. Start of combustion is predicted accurately in each case. The predicted peak pressure is slightly higher for each case and the reason is the combustion efficiency is assumed 100% in simulation. COM and DPM do not consider the effects of low temperature regions, such as crevices. For the DPM, all fourteen zones have high temperatures which result in complete combustion. This can cause prediction deviations of IMEP and thermal efficiency, but the model is still useful as it is able to capture the effects of valve timing. In each of the five cases, DPM predicts the effective compression ratio slightly lower and this causes a discrepancy between DPM and measured cylinder pressure values during compression. The DPM shows a small abrupt pressure increase during compression before the main combustion and this is caused by early heat release due to the

Low Temperature Reactions (LTR). Table 4 compares some more combustion indices i.e. the prediction and measured location of occurrence of the start of combustion and the predicted and measured peak pressure. This table shows that COM and DPM are accurate for control and thermodynamics analysis of HCCI engines. Both COM and DPM are computationally efficient needing 8 ms and 156 sec respectively to simulate an HCCI cycle on a 2.66 GHz Intel PC.

Table 2. Single cylinder research engine specifications [54]

| | |
|-----------------------|---------|
| Bore | 97 mm |
| Stroke | 88.9 mm |
| Compression ratio | 13.9:1 |
| Connecting rod length | 159 mm |
| IVO [bTDC] | 280 |
| IVC [bTDC] | 180 |
| EVO [bTDC] | -180 |
| EVC [bTDC] | -280 |

Table 3. Operating Conditions for Validation

| Case Name | A | B | C | D | E |
|-----------------|------|------|------|------|------|
| EVC [bTDC] | -300 | -330 | -320 | -320 | -300 |
| IVO [bTDC] | 300 | 330 | 290 | 300 | 320 |
| Ω [RPM] | 798 | 799 | 800 | 801 | 803 |
| ϕ | 0.37 | 0.33 | 0.36 | 0.33 | 0.33 |
| T_{int} [C] | 80 | 80 | 80 | 78.9 | 78.4 |
| P_{int} [Bar] | 0.88 | 0.88 | 0.88 | 0.89 | 0.89 |

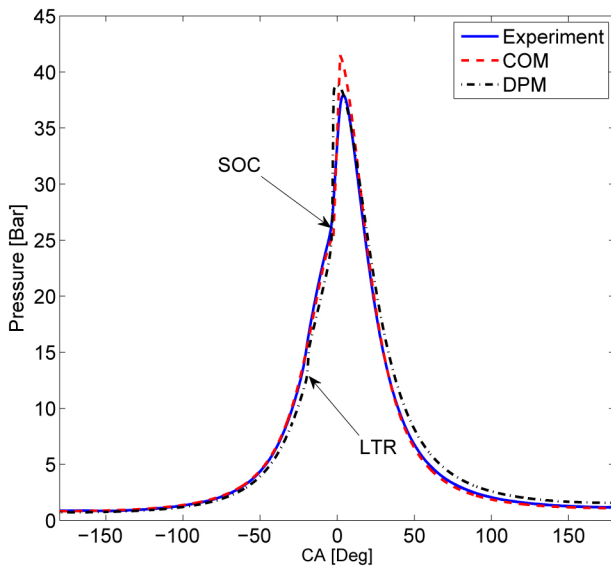


Figure 1. Experiment Pressure Trace vs. Models - Case A (see Table 3)

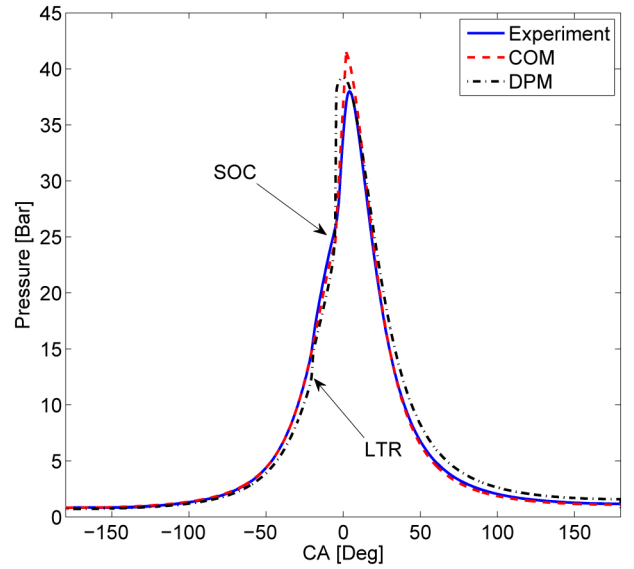


Figure 2. Experiment Pressure Trace vs. Models - Case B (see Table 3)

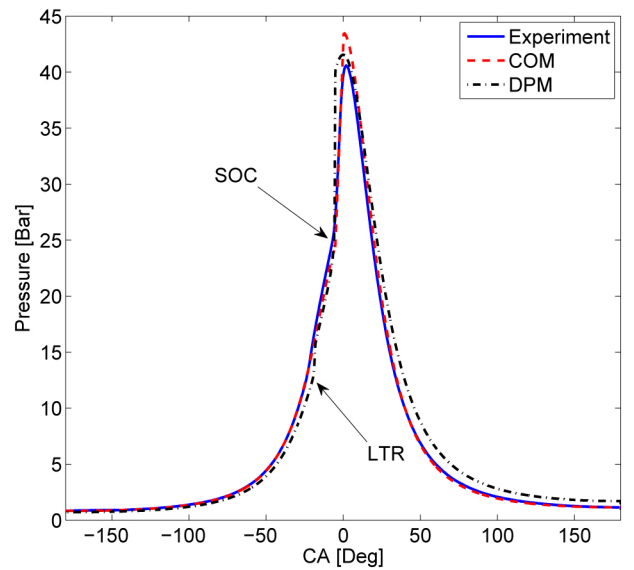


Figure 3. Experiment Pressure Trace vs. Models - Case C (see Table 3)

To further validate both COM and DPM, more experimental data is used. Figure 6 shows predicted and measured θ_{50} when IVO timing changes while holding all other parameters constant. COM and DPM predictions are all within 1-3 crankangle degrees of the measured values. As shown in Figure 6, combustion timing advances when IVO timing is retarded. When IVO is retarded, in-cylinder gas temperature is reduced due to expansion but more fresh charge is inducted into the cylinder due to low in-cylinder pressure at IVO. Mixture composition has key role on HCCI combustion phasing control in this case and combustion advances because the trapped charge fuel equivalence ratio is increased by late IVO. Figure 7 shows predicted and measured θ_{50} when EVC timing changes while keeping other operating parameters

constant. As shown in Figure 7, COM and DPM predictions are acceptable. Combustion timing retards when EVC timing is advanced. When EVC is advanced, in-cylinder gas temperature is increased because more residual gas is trapped but since the pressure is high at IVO part of the residual gas goes into the intake manifold and dilutes the fresh charge. Combustion timing retards because the trapped charge fuel equivalence ratio is reduced by advanced EVC. COM has been further validated in [55].

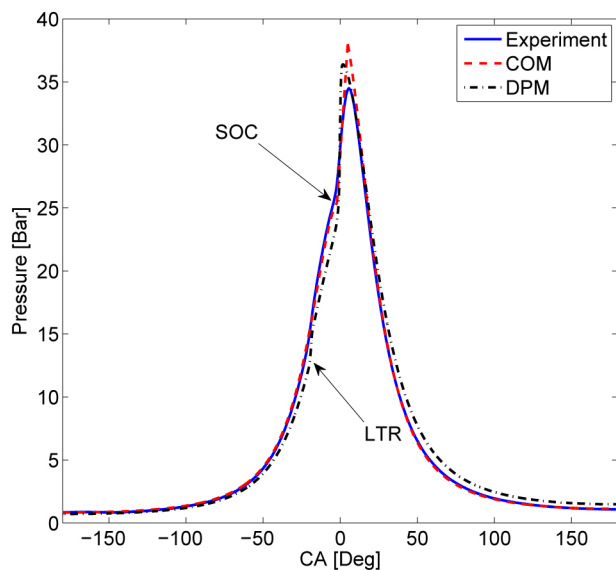


Figure 4. Experiment Pressure Trace vs. Models - Case D (see Table 3)

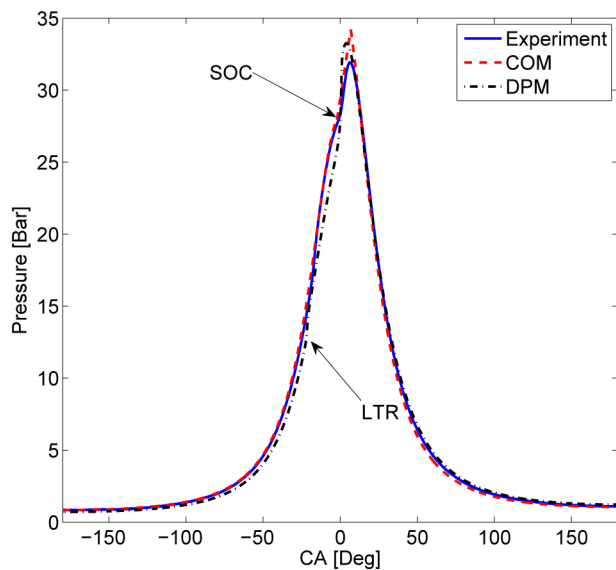
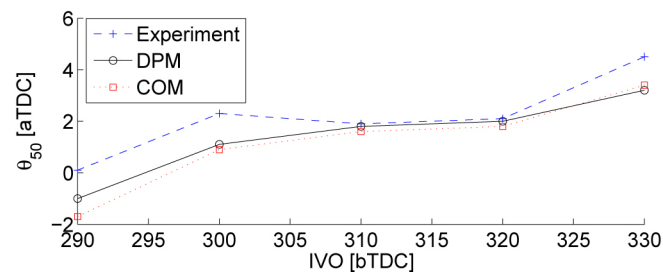


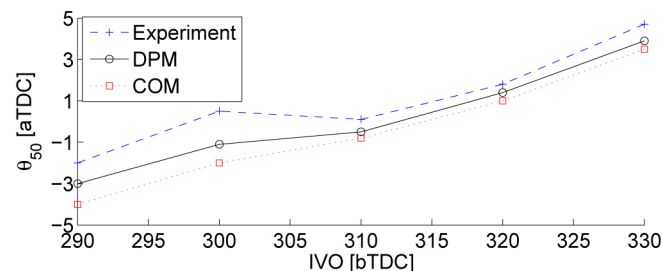
Figure 5. Experiment Pressure Trace vs. Models - Case E (see Table 3)

Table 4. Comparison of predicted and experimental values of peak pressure and start of combustion

| Case Name | A | B | C | D | E |
|---------------------------------------|------|------|------|------|-------|
| Measured SOC [CA aTDC] | -4.3 | -6.2 | -6.5 | -3.7 | -0.1 |
| Measured P_{max} [Bar] | 37.9 | 38 | 40.5 | 34.5 | 31.9 |
| Measured $\theta_{P_{max}}$ [CA aTDC] | 4 | 4 | 1.9 | 5.6 | 6.1 |
| DPM SOC [CA aTDC] | -3.8 | -5.8 | -6.2 | -2.2 | -0.8 |
| DPM P_{max} [Bar] | 38.8 | 39.2 | 41.5 | 36.4 | 33.26 |
| DPM $\theta_{P_{max}}$ [CA aTDC] | 0 | -0.1 | -0.1 | 1.9 | 4 |
| COM SOC [CA aTDC] | -2.4 | -5.1 | -5.2 | -2.1 | -1.9 |
| COM P_{max} [Bar] | 41.5 | 41.6 | 43.5 | 38 | 34.2 |
| COM $\theta_{P_{max}}$ [CA aTDC] | 1.9 | 1.8 | 0.4 | 4.9 | 6.9 |

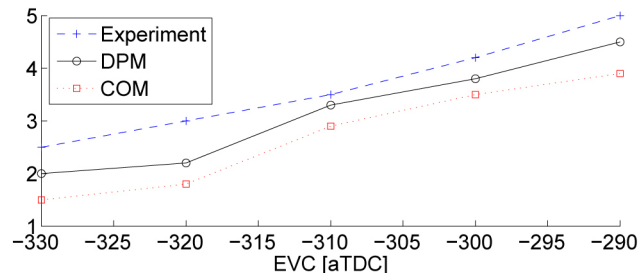


(a). Injected Fuel Energy = 0.36 kJ



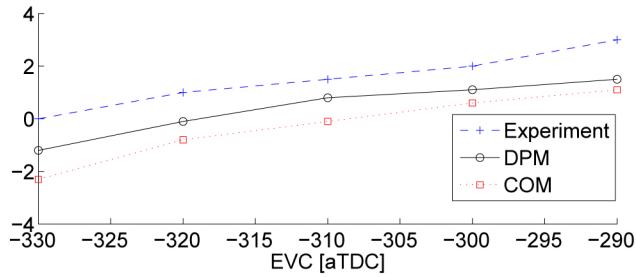
(b). Injected Fuel Energy = 0.39 kJ

Figure 6. Comparison of predicted and measured θ_{50} for IVO timing changes for two injected fuel energy levels (EVC=-320 bTDC, $T_{int}=88^\circ\text{C}$, $P_{int}=0.88$ Bar and $\Omega=819$ RPM)



(a). Injected Fuel Energy = 0.36 kJ

Figure 7. Comparison of predicted and measured θ_{50} for EVC timing changes for two injected fuel energy levels (IVO=320 bTDC, $T_{int}=88^\circ\text{C}$, $P_{int}=0.88$ Bar and $\Omega=819$ RPM)



(b). Injected Fuel Energy = 0.39 kJ

Figure 7. (cont.) Comparison of predicted and measured θ_{50} for EVC timing changes for two injected fuel energy levels ($IVO=320$ bTDC, $T_{int}=88^\circ\text{C}$, $P_{int}=0.88$ Bar and $\Omega=819$ RPM)

IMPLEMENTATION OF A VARIABLE VALVE TIMING STRATEGY WITH COM AND DPM

VVT is a useful technology for development of HCCI engines [56, 57, 58]. HCCI combustion control with internal EGR is one effective way to avoid misfire at low load and knock at high load. The COM and DPM are used to study the effect of variable valve timing on HCCI combustion. Symmetric Negative Valve Overlap (SNVO) is used as a VVT strategy. This strategy involves early closing of the exhaust valves followed by late intake valve opening. Figure 8a shows a typical pressure trace diagram of HCCI engine with SNVO from the DPM. In Figure 8b, the pressure trace around TDC of combustion is scaled to show more clearly important HCCI combustion indices.

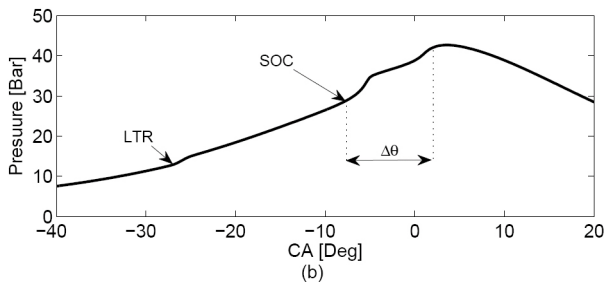
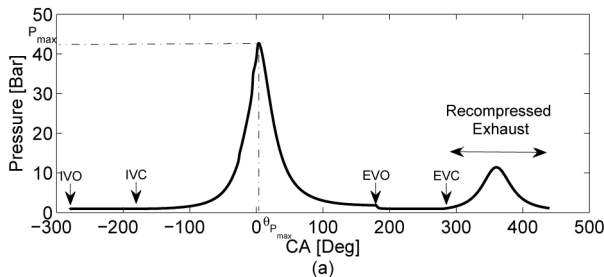


Figure 8. Simulated Pressure Trace of HCCI engine with 160 Deg SNVO ($\Omega=825$ RPM, $\phi=0.3$)

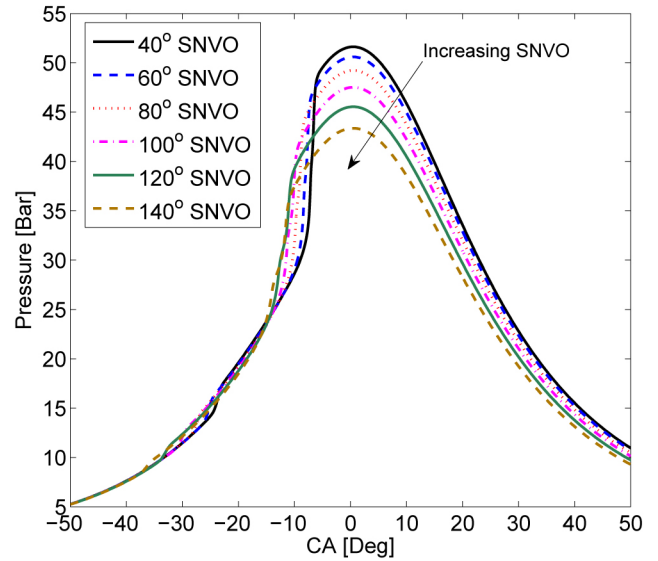


Figure 9. DPM Simulated Pressure Trace of HCCI engine with variable SNVO ($\Omega=825$ RPM)

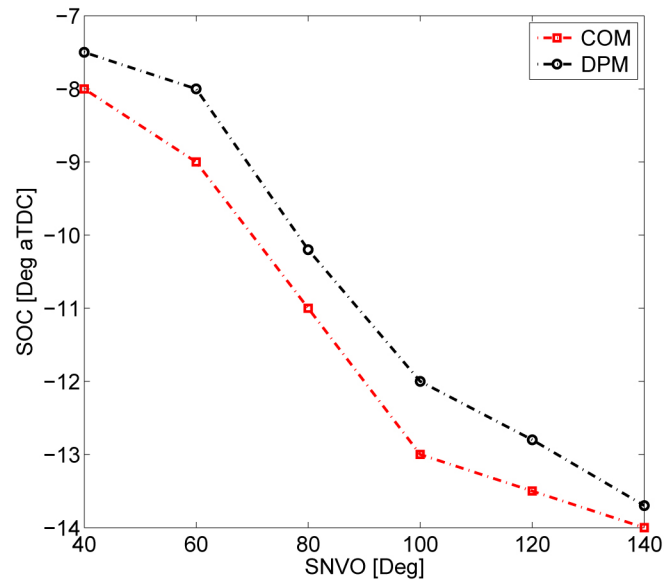


Figure 10. DPM and COM Simulated SOC with variable SNVO ($\Omega=825$ RPM)

As shown in Figure 8a, EVC and IVO timing changes are symmetric around TDC so the recompression work can be regained as expansion work. Six cases are examined, for which SNVO is 40, 60, 80, 100, 120 and 140 degrees. The fueling rate, intake manifold temperature and engine speed are held constant and only the valve timing is changed. DPM and COM simulation results are shown in Figures 9 - 10. Figure 9 shows the in-cylinder pressure from DPM and Figure 10 shows the crank angle of SOC from COM and DPM. When SNVO increases, the amount of internal EGR as well as the in-cylinder gas temperature increases. EGR dilutes the in-cylinder mixture leading to a lower equivalence ratio. SOC is advanced when the in-cylinder gas temperature increases with larger amounts of internal EGR but the peak

pressure is reduced and the combustion duration increases due to lower equivalence ratio of the trapped mixture. These results are consistent with [56].

CONCLUSION

A detailed multi-zone HCCI model (DPM) along with a control oriented model (COM) have been developed and implemented in a full cycle simulation of an HCCI engine for the purpose of predicting and controlling HCCI combustion characteristics. A validation of the both the DPM and COM against experiments in a single cylinder research engine have been conducted over a range of engine loads and valve timings. Comparison of DPM and COM performance with available experimental data shows good agreement. VVT modulates the internal EGR and is one effective way for HCCI combustion timing control. Use of a symmetric negative valve overlap strategy is investigated with both DPM and COM. A wide range of internal EGR can be obtained with this VVT strategy. Ignition timing is advanced and peak in-cylinder pressure is lowered by higher symmetric negative valve overlap durations. In future work, both COM and DPM will be validated against experimental data for other operating ranges.

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NOMENCLATURE

SYMBOLS

| | | |
|----------------|-----------------------------------|----------------------|
| A | Area | $[m^2]$ |
| θ_{50} | Crank angle for 50% burnt fuel | $[CAD \text{ aTDC}]$ |
| C_d | Discharge coefficient | $[-]$ |
| c_v | Constant-volume heat capacity | $[\frac{kJ}{kg.K}]$ |
| c_p | Constant-pressure heat capacity | $[\frac{kJ}{kg.K}]$ |
| $\Delta\theta$ | Combustion duration | $[CAD]$ |
| EGR | Fraction of exh. gas recirculated | $[-]$ |
| m | Mass | $[kg]$ or $[g]$ |
| Ω | Engine speed | $[rpm]$ |
| ϕ | Equivalence ratio | $[-]$ |
| T | Temperature | $[K]$ or $[C]$ |
| θ | Crank angle | $[CAD]$ |
| U | Internal energy | $[kJ]$ |
| V | Volume | $[m^3]$ |
| P | Pressure | $[Bar]$ |
| M | Molecular weight | $[kg]$ |
| W | Displacement work | $[kJ]$ |
| y | Mass fraction | $[-]$ |

ABBREVIATIONS

| | |
|------|----------------------------------|
| aTDC | after Top Dead Center |
| CAD | Crank Angle Degree |
| CFD | Computational Fluid Dynamics |
| CI | Compression Ignition |
| SNVO | Symmetric Negative Valve Overlap |
| DPM | Detailed Physical Model |
| COM | Control Oriented Model |
| LTR | Low Temperature Reactions |

The Engineering Meetings Board has approved this paper for publication. It has successfully completed SAE's peer review process under the supervision of the session organizer. This process requires a minimum of three (3) reviews by industry experts.

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